Environmental Protection Agency

- (b) International Civil Aviation Organization, Document Sales Unit, 999 University Street, Montreal, Quebec, Canada H3C 5H7, (514) 954–8022, www.icao.int, or sales@icao.int.
- (1) Annex 16 to the Convention on International Civil Aviation, Environmental Protection, Volume II—Aircraft Engine Emissions, Third Edition, July 2008 (ICAO Annex 16). IBR approved for §§87.1, 87.42(c), and 87.60(a) and (b).
 - (2) [Reserved]

[77 FR 36381, June 18, 2012]

Subpart B—Engine Fuel Venting Emissions (New and In-Use Aircraft Gas Turbine Engines)

§87.10 Applicability.

- (a) The provisions of this subpart are applicable to all new aircraft gas turbines of classes T3, T8, TSS and TF equal to or greater than 36 kilonewton rated output, manufactured on or after January 1, 1974, and to all in-use aircraft gas turbine engines of classes T3, T8, TSS and TF equal to or greater than 36 kilonewton rated output manufactured after February 1, 1974.
- (b) The provisions of this subpart are also applicable to all new aircraft gas turbines of class TF less than 36 kilonewton rated output and class TP manufactured on or after January 1, 1975 and to all in-use aircraft gas turbines of class TF less than 36 kilonewton rated output and class TP manufactured after January 1, 1975.

[49 FR 41002, Oct. 18, 1984]

§87.11 Standard for fuel venting emissions.

- (a) No fuel venting emissions shall be discharged into the atmosphere from any new or in-use aircraft gas turbine engine subject to the subpart. This paragraph is directed at the elimination of intentional discharge to the atmosphere of fuel drained from fuel nozzle manifolds after engines are shut down and does not apply to normal fuel seepage from shaft seals, joints, and fittings.
- (b) Conformity with the standard set forth in paragraph (a) of this section shall be determined by inspection of

the method designed to eliminate these emissions.

Subpart C—Exhaust Emissions (New Aircraft Gas Turbine Engines)

§87.20 Applicability.

The provisions of this subpart are applicable to all aircraft gas turbine engines of the classes specified beginning on the dates specified.

§ 87.21 Exhaust emission standards for Tier 4 and earlier engines.

This section describes the emission standards that apply for Tier 4 and earlier engines that apply for aircraft engines manufactured before July 18, 2012 and certain engines exempted under $\S87.50$. Note that the tier of standards identified for an engine relates to NO_X emissions and that the specified standards for HC, CO, and smoke emissions apply independent of the changes to the NO_X emission standards.

- (a) Exhaust emissions of smoke from each new aircraft gas turbine engine of class T8 manufactured on or after February 1, 1974, shall not exceed: Smoke number of 30.
- (b) Exhaust emissions of smoke from each new aircraft gas turbine engine of class TF and of rated output of 129 kilonewtons thrust or greater, manufactured on or after January 1, 1976, shall not exceed:

SN=83.6(r0) $^{-0.274}$ (r0 is in kilonewtons).

- (c) Exhaust emission of smoke from each new aircraft gas turbine engine of class T3 manufactured on or after January 1, 1978, shall not exceed: Smoke number of 25.
- (d) Gaseous exhaust emissions from each new commercial aircraft gas turbine engine shall not exceed:
- (1) Classes TF, T3, T8 engines greater than 26.7 kilonewtons rated output:
- (i) Engines manufactured on or after January 1, 1984:

Hydrocarbons: 19.6 grams/kilonewton

(ii) Engines manufactured on or after July 7, 1997.

Carbon Monoxide: 118 grams/kilonewton rO.

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(iii) The following Tier 0 emission standard applies for engines of a type or model of which the date of manufacture of the first individual production model was on or before December 31, 1995 and for which the date of manufacture of the individual engine was on or before December 31, 1999.

Oxides of Nitrogen: (40 + 2(rPR)) grams/kilonewton rO.

(iv) The following Tier 2 emission standard applies for engines of a type or model of which the date of manufacture of the first individual production model was after December 31, 1995 or for which the date of manufacture of the individual engine was after December 31, 1999:

Oxides of Nitrogen: (32 + 1.6(rPR)) grams/kilonewton rO.

- (v) The emission standards prescribed in paragraphs (d)(1) (iii) and (iv) of this section apply as prescribed beginning July 7, 1997.
- (vi) The following Tier 4 emission standards apply for engines of a type or model of which the date of manufacture of the first individual production model was after December 31, 2003:
- (A) Engines with a rated pressure ratio of 30 or less:
- (1) Engines with a maximum rated output greater than 89 kilonewtons:

Oxides of Nitrogen: (19 + 1.6(rPR)) grams/kilonewtons rO.

(2) Engines with a maximum rated output greater than 26.7 kilonewtons but not greater than 89 kilonewtons:

Oxides of Nitrogen: (37.572 + 1.6(rPR) - 0.2087(rO)) grams/kilonewtons rO.

(B) Engines with a rated pressure ratio greater than 30 but less than 62.5:

(1) Engines with a maximum rated output greater than 89 kilonewtons:

Oxides of Nitrogen: (7 + 2(rPR)) grams/kilonewtons rO.

(2) Engines with a maximum rated output greater than 26.7 kilonewtons but not greater than 89 kilonewtons:

Oxides of Nitrogen: $(42.71 + 1.4286(rPR) - 0.4013(rO) + 0.00642(rPR \times rO))$ grams/kilonewtons rO.

(C) Engines with a rated pressure ratio of 62.5 or more:

Oxides of Nitrogen: (32 + 1.6(rPR)) grams/kilonewtons rO.

(vii) The emission standards prescribed in paragraph (d)(1)(vi) of this section shall apply as prescribed beginning December 19, 2005.

(2) Class TSS: Engines manufactured on or after January 1, 1984:

Hydrocarbons=140(0.92)^{rPR} grams/kilonewtons rO.

- (e) Smoke exhaust emissions from each gas turbine engine of the classes specified below shall not exceed:
- (1) Class TF of rated output less than 26.7 kilonewtons manufactured on or after August 9, 1985:
- SN = 83.6(rO) $^{-0.274}$ (rO is in kilonewtons) not to exceed a maximum of SN = 50.
- (2) Classes T3, T8, TSS and TF of rated output equal to or greater than 26.7 kilonewtons manufactured on or after January 1, 1984:

SN=83.6(ro)^{-0.274} (ro is in kilonewtons) not to exceed a maximum of SN=50.

(3) Class TP of rated output equal to or greater than 1,000 kilowatts manufactured on or after January 1, 1984:

 $SN=187(ro)^{-0.168}$ (ro is in kilowatts)

(f) The standards in this section refer to a composite emission sample measured and calculated in accordance with the procedures described in subpart G of this part.

 $[47\ \mathrm{FR}\ 58470,\ \mathrm{Dec.}\ 30,\ 1982,\ \mathrm{as}\ \mathrm{amended}\ \mathrm{at}\ 49\ \mathrm{FR}\ 31875,\ \mathrm{Aug.}\ 9,\ 1984;\ 62\ \mathrm{FR}\ 25365,\ \mathrm{May}\ 8,\ 1997;\ 70\ \mathrm{FR}\ 69686,\ \mathrm{Nov.}\ 17,\ 2005;\ 77\ \mathrm{FR}\ 36381,\ \mathrm{June}\ 18,\ 2012]$

§87.23 Exhaust emission standards for Tier 6 and Tier 8 engines.

This section describes the emission standards that apply for Tier 6 and Tier 8 engines. The standards of this section apply for aircraft engines manufactured on or after July 18, 2012, except where we specify that they apply differently by year, or where the engine is exempt from one or more standards of this section. Except as specified in paragraph (d) of this section, these standards apply based on the date the engine is manufactured. Where a gaseous emission standard is specified by a formula, calculate and round the standard to three significant figures or to the nearest 0.1 g/kN (for standards at or above 100 g/kN). Where a smoke standard is specified by a formula, calculate and round the standard to the